

## ON THE IDENTIFICATION OF THE EGGSHELL ELASTIC PROPERTIES UNDER QUASISTATIC COMPRESSION

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### Abstract

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The problem of the identification of the elastic properties of eggshell, i.e. the evaluation of the Young's modulus and Poisson's ratio is solved. The eggshell is considered as a rotational shell. The experiments on the egg compression under quasistatic loading have been conducted. During these experiments a strain on the eggshell surface has been recorded. By the mutual comparison between experimental and theoretical values of strains the influence of the elastic constants has been demonstrated.

eggshell, elastic shell theory, strain, elastic constants, compression

The investigation of eggs behaviour under compressive loading is a subject of the increasing interest because this loading corresponds to the loading of eggs during their packing. In order to minimize the eggs laid. In order to find all factors which affect the strength of eggshell, it is necessary to determine the strength of the eggshell. In quasistatic compression test, eggs are compressed between two parallel plates by a steadily increasing load until failure results. This procedure enables to determine the force and the shortening of the egg at the moment of failure. These quantities may be affected by the egg shape, by the thickness of the eggshell and by many other factors which are summarized by Bain (1997). The exact evaluation should lead to the values of stress and strain at the failure. These parameters may be easily obtained from test mentioned above if we use flat testing specimens. The egg have a complicated shape. The egg may be considered as a shell of revolution. The stress and strain mentioned above can be determined only by the indirect way. There are some papers dealing with this problem – see e.g. Rehugler (1963), Maneau and Henderson (1970), Tung et al. (1969).

In the given paper we have focused on the solution of this problem using of an advanced elastic shell theo-

ry developed by Gong (1995). This theory enables to obtain values of strain and stress even for the layered shell and for static as well as for impact loading (Shim et al. 1996). At the same time the experiments on the hen's eggs have been performed. The comparison of the results of these experiments with the conclusions of the shell theory led to the possibility of the elastic constants evaluation.

### THEORETICAL BACKGROUND

Figure 1 shows a laminated shell of the axial symmetry. The model enables to describe the orthotropic elasticity of the single shell layer. The meridional and circumferential radii of curvature are denoted by  $R$  and  $r$ , respectively, while the distance from the axis of rotation to the centre of a meridional arc is denoted by  $e$ . The shape of the egg has been determined by Bartoň and Krivánek (2001) as:

$$x = a \cos(\varphi), y = a \sin(\varphi),$$

where

$$a = k_1 \frac{1 + k_2 \cos(\varphi)}{\sqrt{k_3 \cos^2(\varphi) + \sin^2(\varphi)}}$$

The used values of constants are:  $k_1 = 21.079$  mm,  $k_2 = 0.172$ ,  $k_3 = 0.505$ .

For an shell subjected to a distributed transverse load  $q_n$ , the equation of motion can be expressed as (Gong, 1995)

$$\begin{aligned} \frac{\partial N_1}{R \partial \psi} + \frac{\partial N_6}{r \partial \theta} + \frac{Q_1}{R} - \frac{N_1 - N_2}{r} \sin \psi &= I_1 \ddot{u} + I_2 \ddot{\beta}_1 \\ \frac{\partial N_6}{R \partial \psi} + \frac{\partial N_2}{r \partial \theta} + \frac{Q_2}{r} \cos \psi - \frac{2N_6}{r} \sin \psi &= I_1 \ddot{v} + I_2 \ddot{\beta}_2 \\ \frac{\partial Q_1}{R \partial \psi} + \frac{\partial Q_2}{r \partial \theta} - \frac{N_1}{R} - \frac{N_2}{r} \cos \psi - \frac{Q_1}{r} \cos \psi &= \rho h \ddot{w} - q_n \\ \frac{\partial M_1}{R \partial \psi} + \frac{\partial M_6}{r \partial \theta} - Q_1 - \frac{M_1 - M_2}{r} \sin \psi &= I_2 \\ \frac{\partial M_6}{R \partial \psi} + \frac{\partial M_2}{r \partial \theta} - Q_2 - \frac{2M_6}{r} \sin \psi &= I_2 \ddot{v} + I_3 \ddot{\beta}_2 \end{aligned}$$

where  $u, v$  and  $w$  are the displacements along the meridional  $\psi$  – axis, circumferential  $\theta$  – axis and radial  $\zeta$  – axis;  $\beta_1$  and  $\beta_2$  are bending slopes in the  $\psi$ – $\zeta$  and  $\theta$ – $\zeta$  planes;  $I_i$  are inertia factors defined by Gong (1995) as:

$$(I_1, I_2, I_3) = \sum_{k=1}^{L_k} \int_{z_{k-1}}^{z_k} \rho^{(k)}(1, z, z^{(2)}) dz,$$

where  $L_k$  is the number of single layers in the shell.

$N_i, M_i, Q_1$  and  $Q_2$  are stress resultants given by:

$$(N_i, M_i) = \sum_{k=1}^{L_k} \int_{\zeta_{k-1}}^{\zeta_k} \sigma_i^{(k)} d\zeta \quad (i = 1, 2, 6)$$

$$(Q_1, Q_2) = \sum_{k=1}^{L_k} \int_{\zeta_{k-1}}^{\zeta_k} (\sigma^{(3)} \sigma^{(4)}) d\zeta.$$

The relationship between the stress resultants and the strains induced are described by the following equations:

$$\begin{aligned} N_i &= A_{ij} \varepsilon_j^0 + B_{ij} k_j^0 \\ M_i &= B_{ij} \varepsilon_j^0 + D_{ij} k_j^0 \quad (i, j = 1, 2, 6) \\ Q_1 &= A_{5j} \varepsilon_j^0 \end{aligned}$$

$$Q_2 = A_{4j} \varepsilon_j^0,$$

where

$$(A_{ij}, B_{ij}, D_{ij}) = \sum_{k=1}^{L_k} \int_{\zeta_{k-1}}^{\zeta_k} D_{ij}^k(1, \zeta, \zeta^2) d\zeta.$$

Strains associated with the deflections and rotations can also be expressed as function of  $\psi$  and  $\theta$ :

$$\begin{aligned} \varepsilon_1 &= \varepsilon_1^0 + \zeta k_1^0 & \varepsilon_2 &= \varepsilon_2^0 + \zeta k_2^0 \\ \varepsilon_3 &= 0 & \varepsilon_4 &= \varepsilon_4^0 \\ \varepsilon_5 &= \varepsilon_5^0 & \varepsilon_6 &= \varepsilon_6^0 + \zeta k_6^0, \end{aligned}$$

where

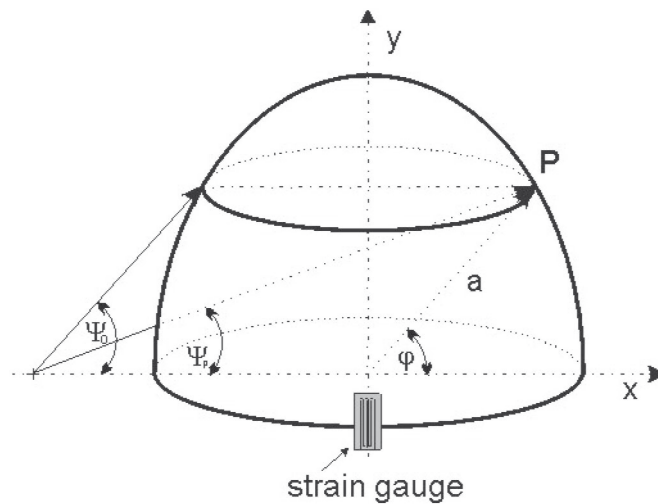
$$\begin{aligned} \varepsilon_1^0 &= \frac{1}{R} \frac{\partial u}{\partial \psi} + \frac{w}{R} \\ \varepsilon_2^0 &= \frac{1}{r} \frac{\partial v}{\partial \theta} + \frac{w}{r} \cos \psi - \frac{u}{r} \sin \psi \\ \varepsilon_4^0 &= \frac{1}{r} \frac{\partial w}{\partial \theta} + \beta_2 - \frac{v}{r} \cos \psi \\ \varepsilon_5^0 &= \frac{1}{R} \frac{\partial v}{\partial \psi} + \frac{1}{r} \frac{\partial u}{\partial \theta} + \frac{v}{r} \sin \psi \\ k_1^0 &= \frac{1}{R} \frac{\partial \beta_1}{\partial \psi} \\ k_2^0 &= \frac{1}{r} \frac{\partial \beta_2}{\partial \theta} - \frac{\beta_1}{r} \sin \psi \\ k_3^0 &= \frac{1}{R} \frac{\partial \beta_2}{\partial \psi} + \frac{1}{r} \frac{\partial \beta_1}{\partial \theta} + \frac{\beta_2}{r} \sin \psi. \end{aligned}$$

By substituting of single equations we obtain a final form of motion of a shell in terms of deflections and rotations:

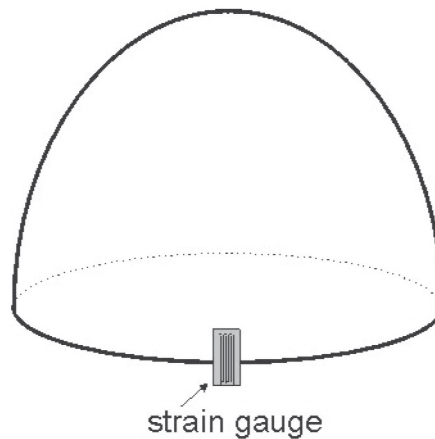
$$[L_{ij}][u, v, w, \beta_1, \beta_2] = \{0, 0, \rho h \ddot{w} - q_n, 0, 0\},$$

where  $[ ]$  and  $\{ \}$  represents matrices and vectors, respectively and  $L_{ij}$  are the differential operators.

These solution of these equations can be found e.g. in paper (Shim et al., 1996).



1: Schematic of the eggshell and the used geometry



2: Detail of the strain gauge position. The length of the strain gauge is 2 mm. The strain has been recorded via the "Spider" recorder

## EXPERIMENTAL RESULTS

For the experiments hen's eggs have been used. These eggs have been compressed using of the testing machine TIRATEST at the velocity 20 mm per minute. On the surface of the egg a strain gauge has been glued see Fig. 2. The egg has been loaded by the following way:

- Loading up to 10 N and subsequent unloading
- The previous step is repeated for the forces 15, 20, 25, 30 N
- The egg has been loaded up to failure.

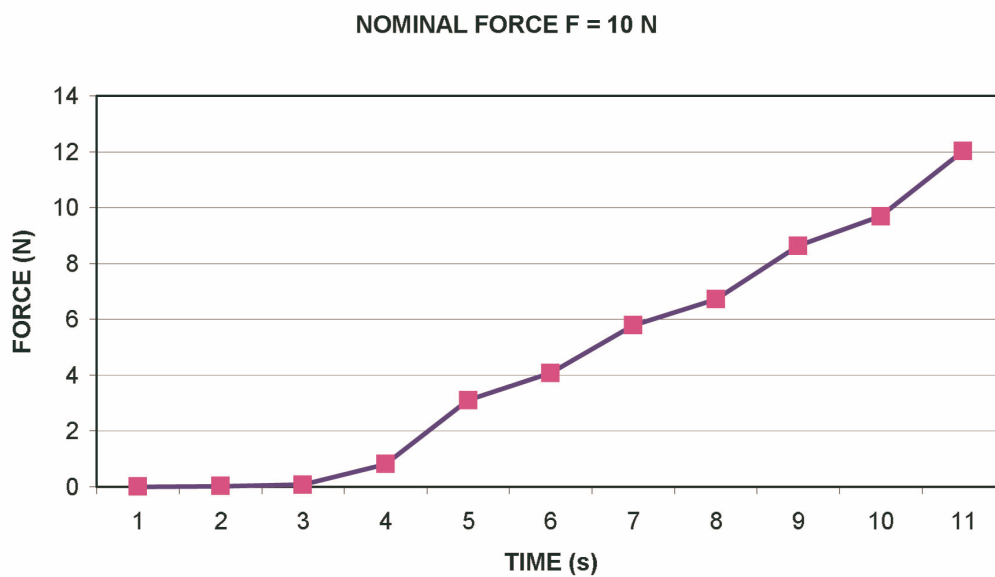
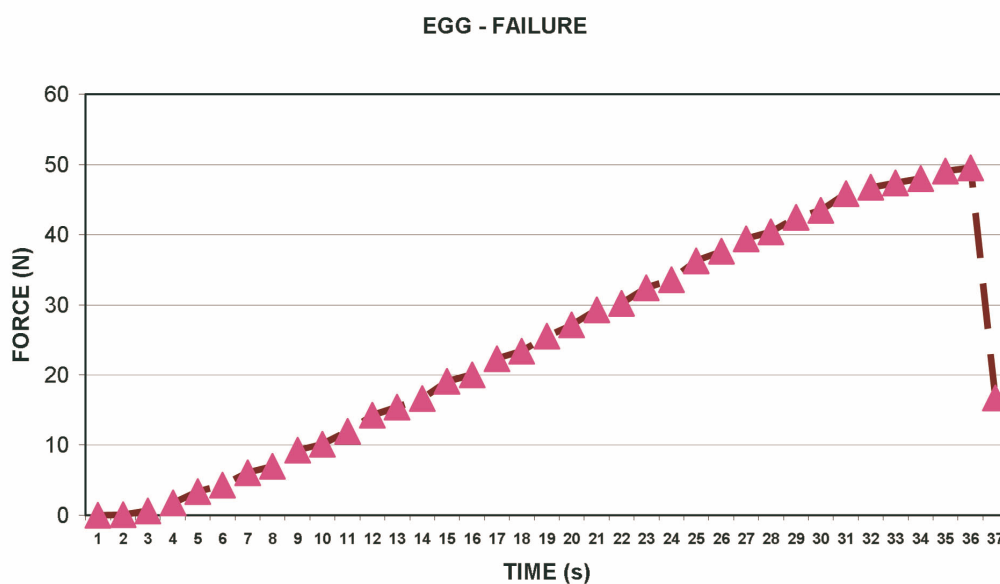
In Fig. 3 the record of time dependence of the force

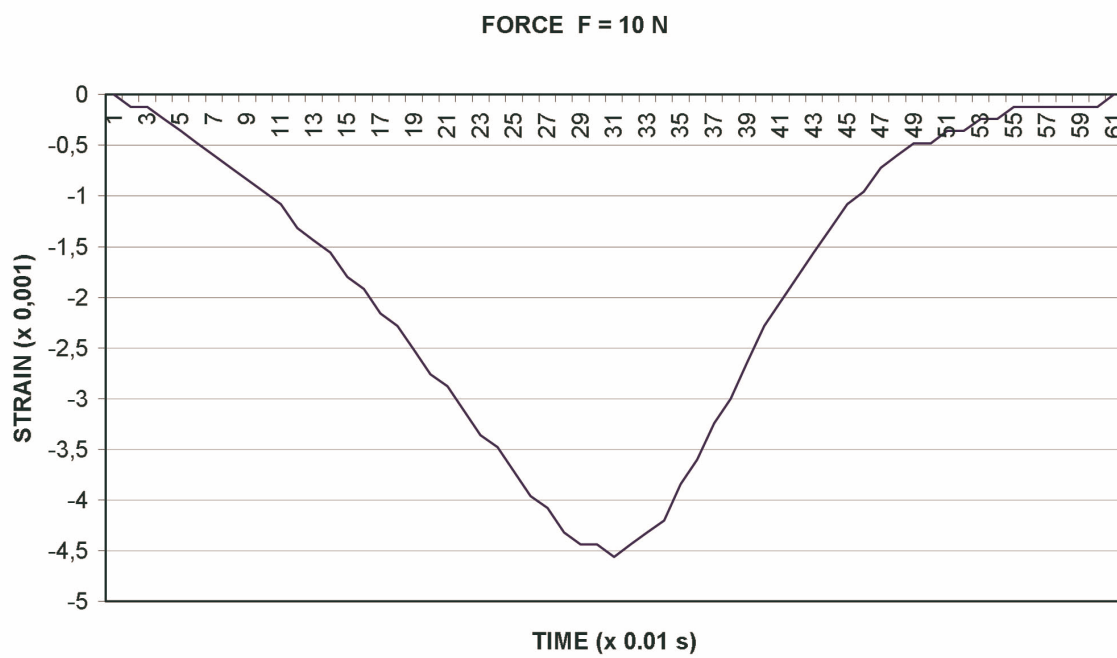
at the top of egg is displayed. The fig. 4. shows the record of the force in the case of the egg failure. The force at the failure is about 49 N. The failure has a form of a thin crack propagation. The record of the strain is displayed in Fig. 5. The record in Fig. 3 corresponds to the loading to about 12 N. The egg deforms only elastically. The time record of strain in the case of the egg failure is shown in Fig. 6. One may see the release of the strain energy during the failure. In table I the values of maximum strains at the maximum load force  $F$  are given. At the same time the values of the egg shortening are also reported. The strain is increasing function of the maximum loading force as shown in Fig. 7. The dependence is linear.

1: *Experimental values of the loading force, strain and egg shortening*

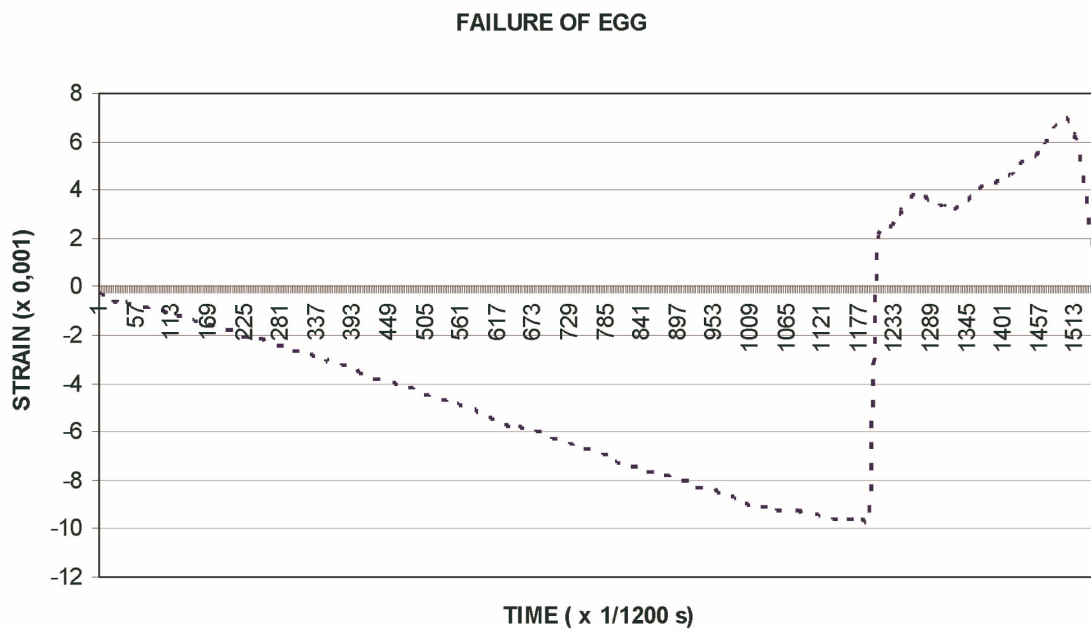
FORCE F (N)	12.03	15.26	20.61	25.83	30.91	49.54
STRAIN (x 0.001)	4.32	4.68	5.76	6.96	7.68	9.60
SHORTENING (mm)	0.100	0.127	0.133	0.187	0.222	0.347

The shortening of the egg depends on the loading force linearly as it is documented in Fig. 8. The linearity of this dependence should be verified for another points on the surface of the egg.

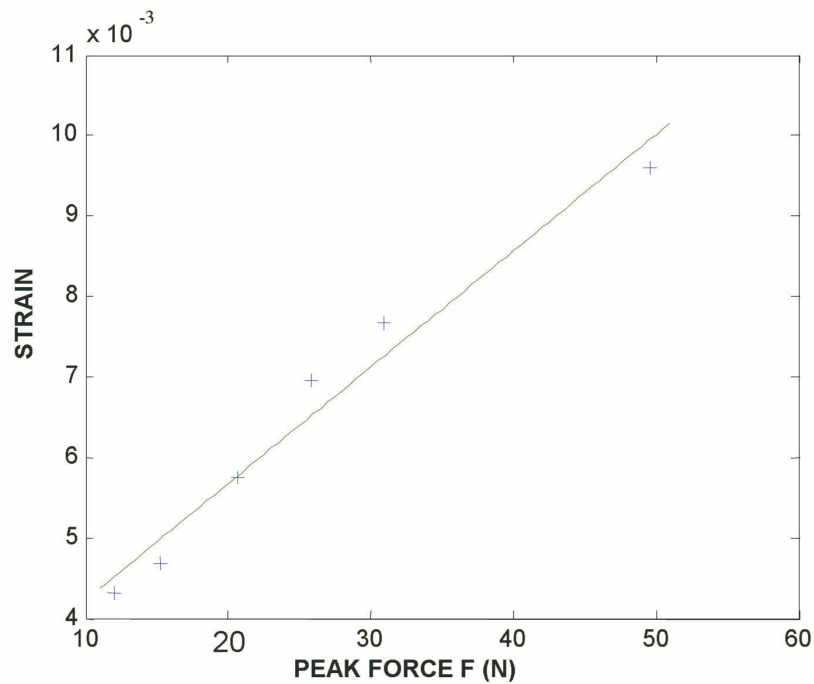
3: *The experimental record loading force on the time. No damage of the egg occurred.*4: *The experimental record loading force on the time. At the maximum of the force failure of the egg starts. The failure is in the form of the single crack. The crack propagates nearly along the meridian.*



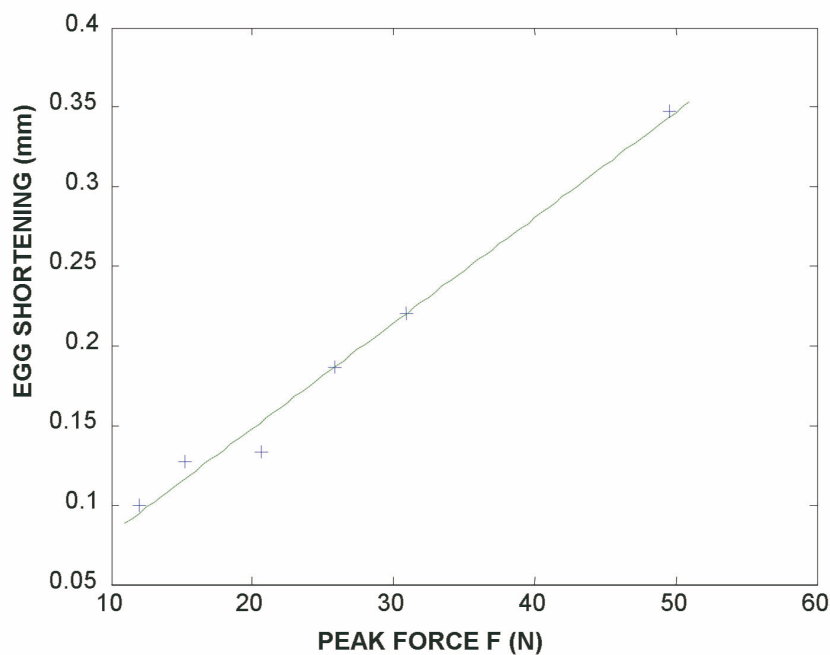
5: The experimental record of the strain



6: The experimental record of the strain. The egg has been failed.



7: The dependence of the strain on the loading force



8: The change of the egg size as function of the loading force

#### THEORETICAL SOLUTION

In the next step we have used the model of the shell loading described in the Chapter 2. The following values of parameters describing the shape of the egg has been used:

$$k_1 = 21.079 \text{ mm}, k_2 = 0.172, k_3 = 0.505.$$

The eggshell has been taken as homogeneous and isotropic. The thickness of the eggshell is 0.35 mm. The elastic properties of the shell are described by the Young's modulus  $E$  and by the Poisson's ratio  $\nu$ . The values of these parameters are given e.g. in review paper (Bain,1997). According to the different sour-

es the values of  $E$  and  $\nu$  vary in a broad interval. We have performed a series of computations of the strain values for the given values of the loading force and

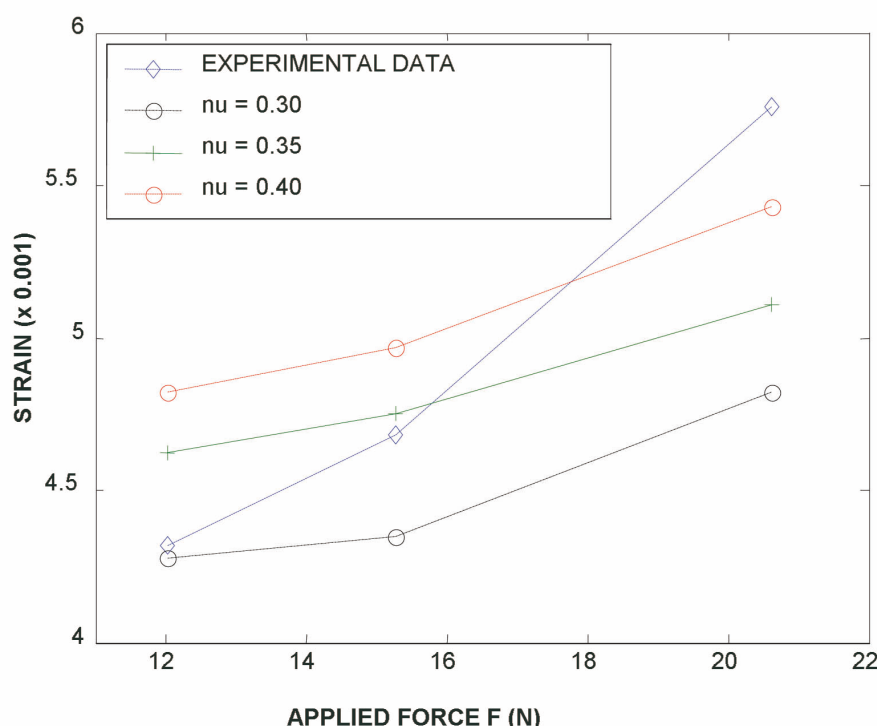
for the different values of  $E$  and  $\nu$ . Results are given in Table II.

## II: Computed values of strains

FORCE (N)	E (Mpa)	$\nu$ (1)	STRAIN (x 0.001)
F = 12.03 N	15 000	0.30	4.28
		0.35	4.62
		0.40	4.82
	25 000	0.30	3.96
		0.35	4.15
		0.40	4.26
	35 000	0.30	3.27
		0.35	3.56
		0.40	3.98
F = 15.26 N	15 000	0.30	4.35
		0.35	4.75
		0.40	4.97
	25 000	0.30	4.18
		0.35	4.42
		0.40	4.75
	35 000	0.30	4.03
		0.35	4.27
		0.40	4.51
F = 20.61 N	15 000	0.30	4.82
		0.35	5.11
		0.40	5.43
	25 000	0.30	4.61
		0.35	4.93
		0.40	5.20
	35 000	0.30	4.43
		0.35	4.78
		0.40	5.07

In Fig. 9 the dependence of the computed value of strain on the applied force is plotted. The Young's

modulus is  $E = 15\,000$  MPa. It is obvious that the strain increases with the value of the Poisson's ratio.



9: The dependence of the strain on the applied force .  $\nu$  denotes the Poisson's ratio. Young's modulus  $E = 15\,000\text{ Mpa}$

In the given figure the experimental values of the strain are also plotted. Very similar results have been obtained for the remaining values of  $E$ . The results indicate that the agreement between the used shell theory and experiments is rather poor. The source of the disagreement may consist in the neglecting of the liquid in the eggs and in some possible error of the strain gauges. The used shell theory also neglects some radial distribution of the stress.

### CONCLUSIONS

In the given paper we have focused on the identi-

fication of the elastic properties of the eggshell. The proposed procedure uses both theory and experiments. The shell theory which has been verified for layered orthotropic shells under static as well as dynamic loading is not too convenient for the description of the egg behaviour under static compression.

The more detail description of the egg deformation is probably needed. This description should involve a liquid behaviour. This situation cannot be considered in framework of the analytical considerations. Some numerical code should be employed. This research will be content of our forthcoming papers.

### SOUHRN

#### O identifikaci elastických vlastností vaječných skořápek při kvazi statickém stlačování

Práce obsahuje úvodní studii zaměřenou na hodnocení elastických konstant vaječné skořápky při stlačování vejce mezi dvěma rovinnými deskami. Postup vychází z výpočtu deformace na povrchu stlačované vaječné skořápky a následného srovnání vypočtených hodnot s experimentálními. V dané práci jsme provedli měření těchto deformací v jednom bodě při různých úrovních zatěžující síly. Z experimentálních dat vyplývá, že jak deformace, tak velikost stlačení závisí na zatěžující síle lineárně, a to až do okamžiku porušení skořápky. Pro výpočet byla použita relativně spolehlivá teorie skořepiny, které byla ověřena jak pro statické, tak zejména pro dynamické zatěžování. Tato teorie umožňuje získat analytické řešení, a to i pro případ vrstevnaté skořepiny s ortotropními elastickými vlastnostmi. Porovnání experimentu a teorie vede k závěru, že použitá teorie nevede k možnosti získat exaktní hodnoty elastických



vlastností. Hodnoty Youngova modulu a Poissonovy konstanty mohou být určeny jen přibližně s tím, že pro jiné hodnoty zátěžné síly může dojít ke zkreslení velikosti napětí. Přesto bude nutné tuto teorii ověřit s tím, že budou provedena měření v dalších bodech a též v dalším směru. Obecně se však ukazuje, že pro detailnější řešení bude vyžadovat numerickou simulaci.

vaječná skořápka, elastické vlastnosti, teorie skořepin, tlakové zatěžování, deformace, lom

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